

Passive EEE parts in the Biomass Payload: challenges and lessons learned

Adriano Carbone, Léo Farhat, Paloma Villar

A. Carbone,

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Outline



- Biomass mission, its objectives and its satellite
- Biomass Payload
- Payload calibration and the importance of the passive elements
- Multipaction: a special topic
- Some other passive EEE topics
- Mission status



*take stock of the biomass in the
world's forests
and
monitor its evolution*

biomass

→ **ESA'S FOREST MISSION**



*take stock of the biomass in the world's forests
and
monitor its evolution*

Primary objectives:

- forest biomass
- forest height
- vegetation disturbances and re-growth

Secondary objectives:

- imaging of sub-surface geology in deserts
- mapping the topography under dense vegetation
- measurements of glacier and ice sheet velocities



*take stock of the biomass in the world's forests
and
monitor its evolution*

Primary objectives:

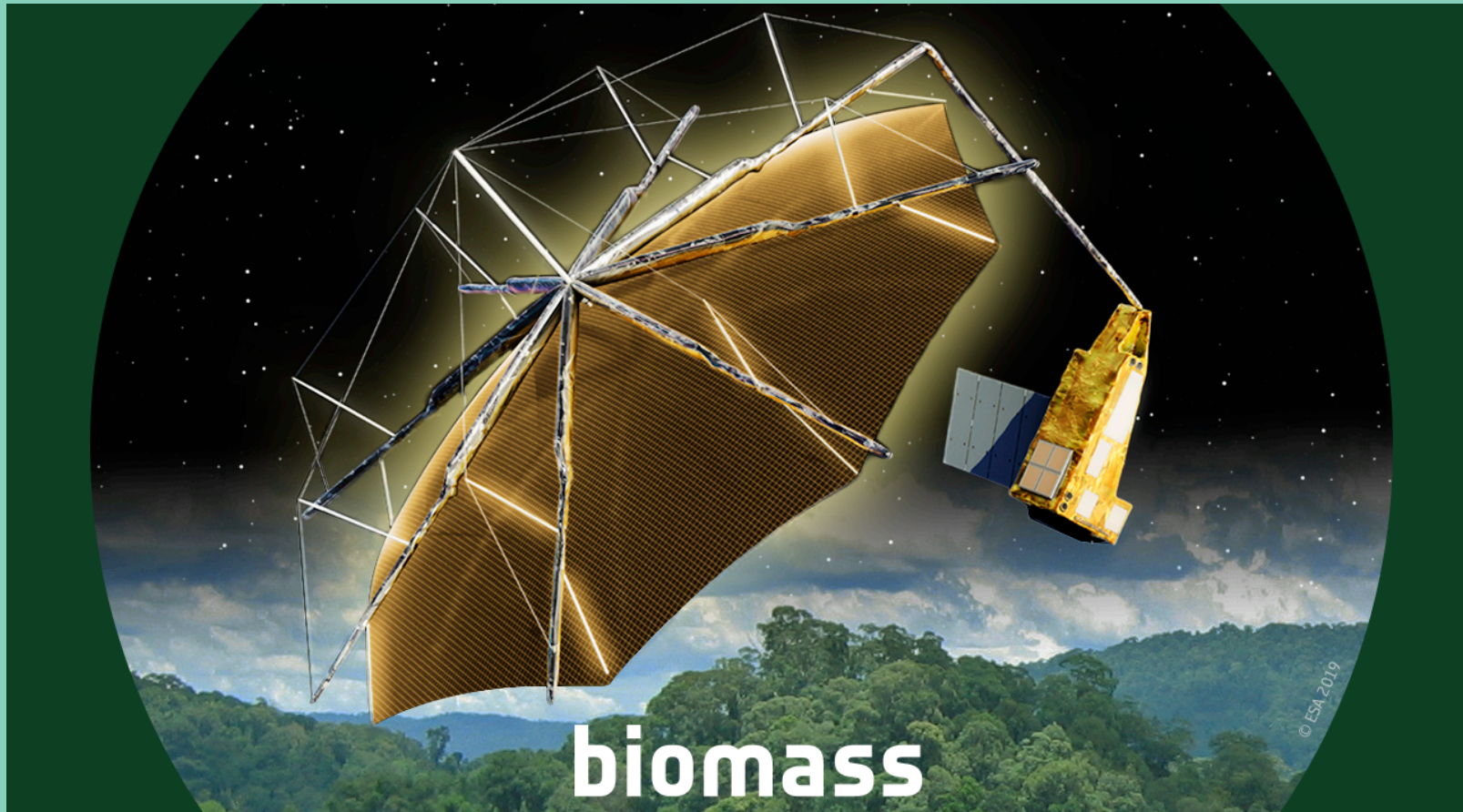
- forest biomass
- forest height
- vegetation disturbances and re-growth

for more details on mission objectives and system architecture see references in last slide

Secondary objectives:

- imaging of sub-surface geology in deserts
- mapping the topography under dense vegetation
- measurements of glacier and ice sheet velocities

Biomass: Space segment



Biomass: Space segment

orbit

dawn-dusk, Sun-synchronous orbit:

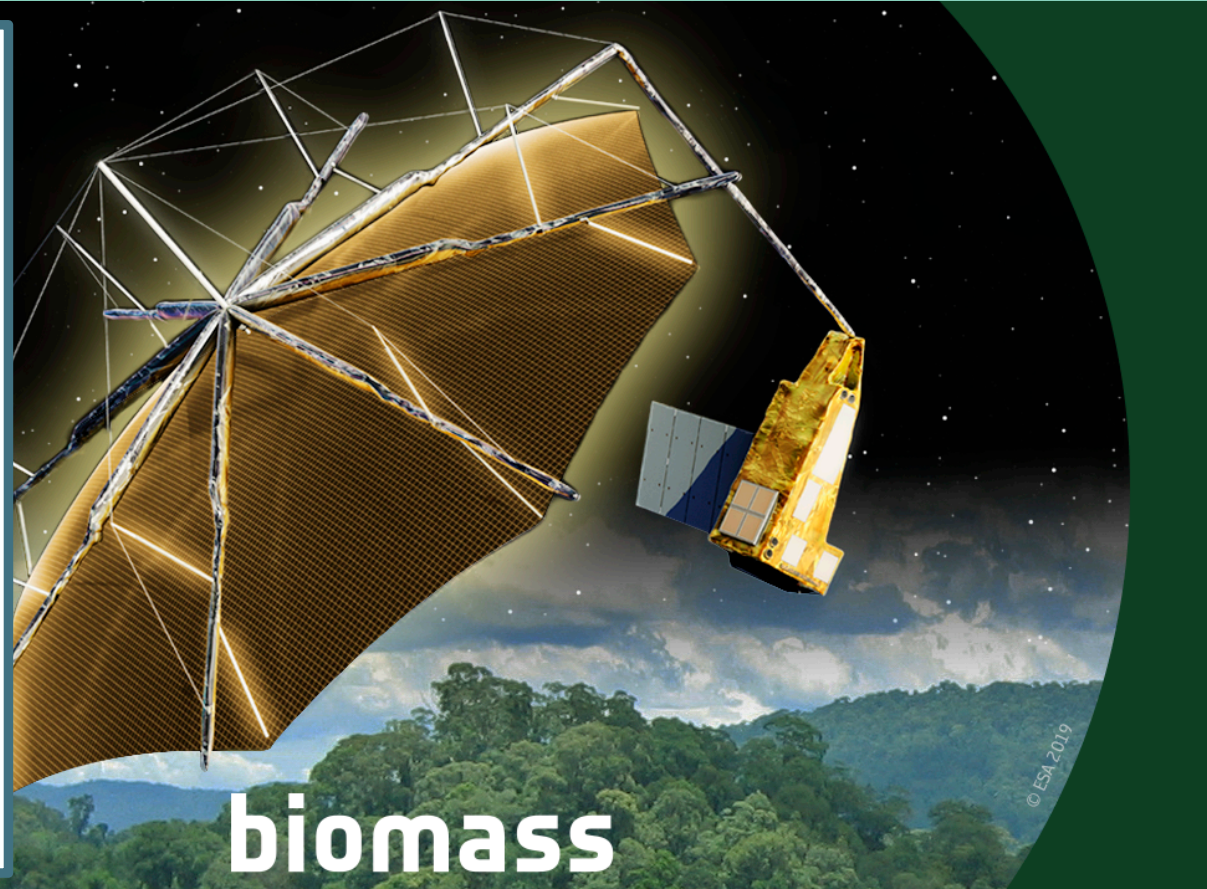
- mean altitude of 666 km,
- inclination $\sim 98^\circ$,
- 3 day repeat cycle orbit (drifting)

spacecraft

single satellite
based on AirbusDS AS250
 ~ 1250 kg
large deployable reflector

payload

P-band (435 ± 3 MHz),
full-polarimetric,
Synthetic Aperture Radar



Biomass: Payload

orbit

dawn-dusk, Sun-synchronous orbit:

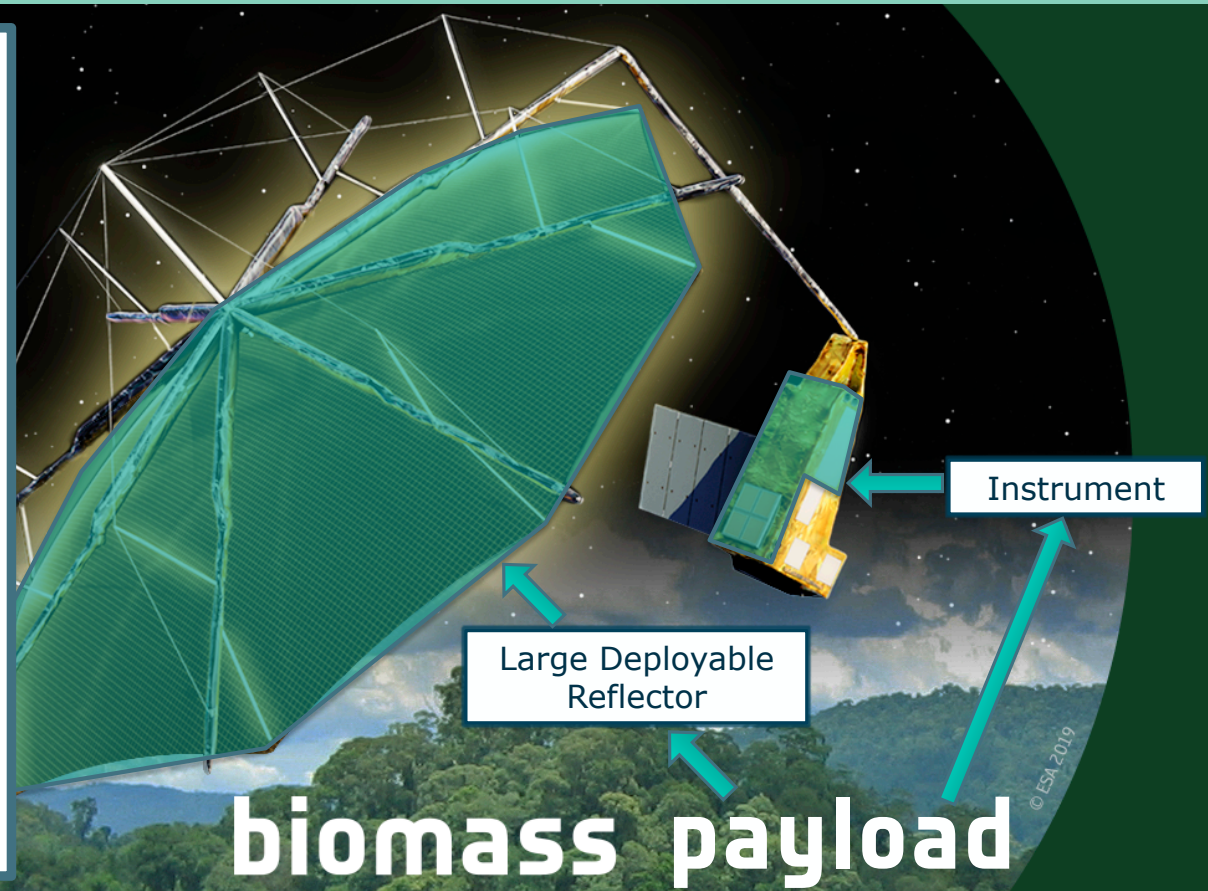
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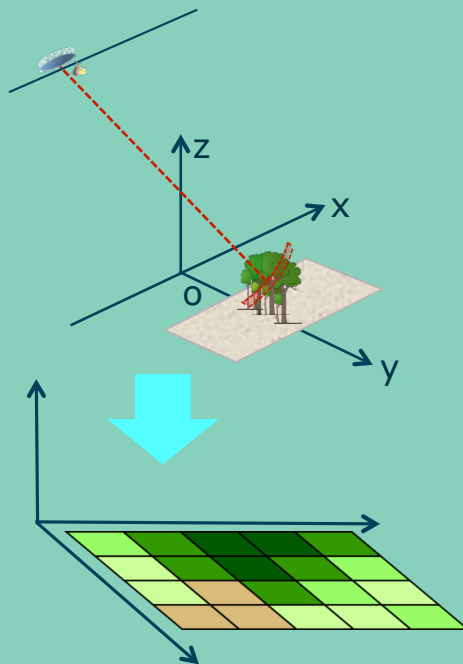
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P-band (435 ± 3 MHz),
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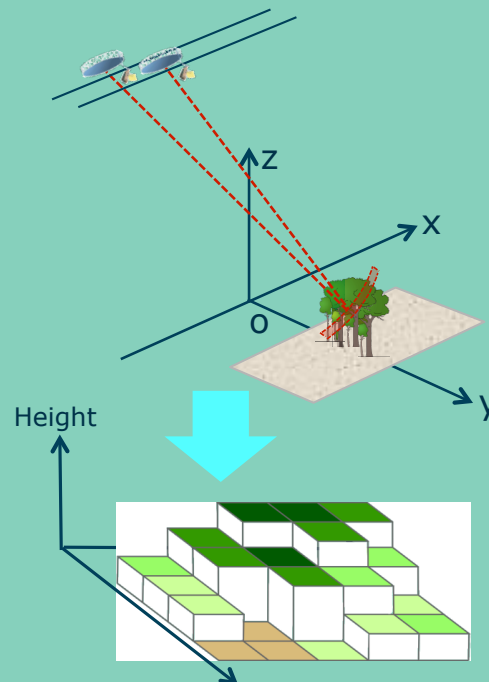


Biomass: observation techniques

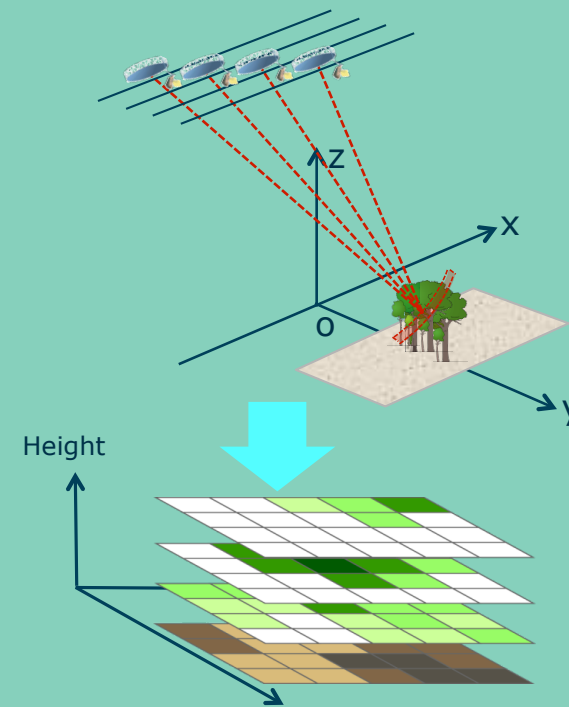
PoISAR
(SAR Polarimetry)



PolInSAR
(Polarimetric SAR Interferometry)



TomoSAR
(SAR Tomography)



Biomass: observation techniques



here goes video

biomass

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Biomass: Payload

orbit

dawn-dusk, Sun-synchronous orbit:

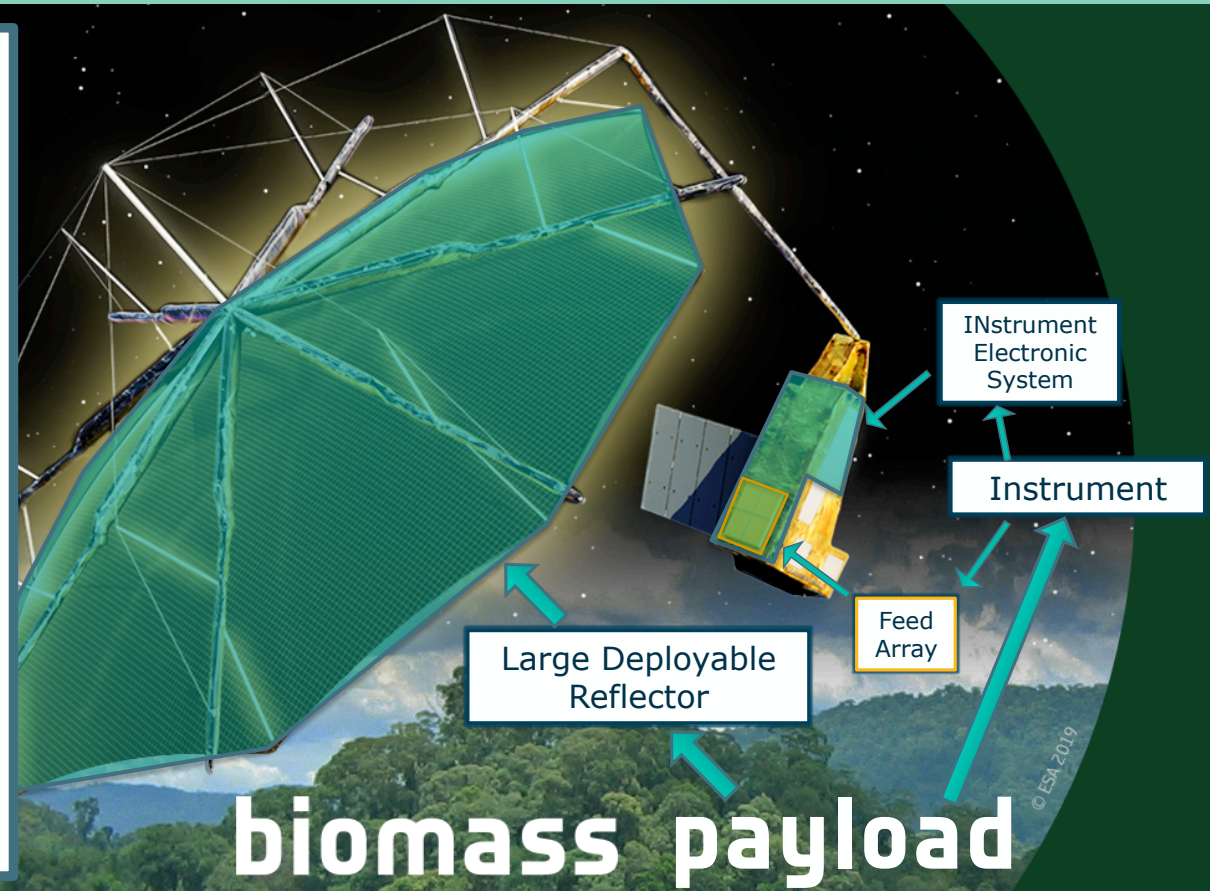
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payload

P-band (435 ± 3 MHz),
full-polarimetric,
Synthetic Aperture Radar



Biomass SAR Instrument: Nomenclature and breakdown

Payload

Instrument

INstrument Electronics System - INES

Digital Control Unit
- DCU

Calibration and Distribution Network
- CDN

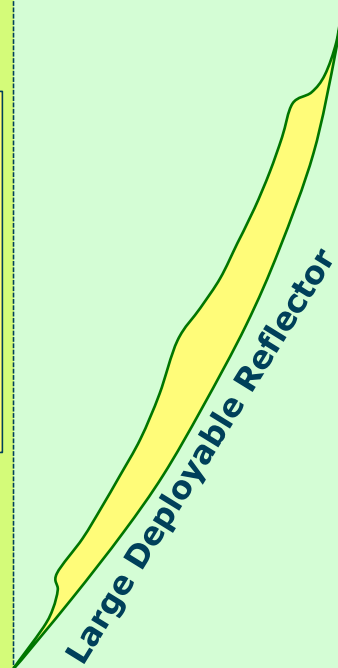
Power Amplifier Subsystem
- PAS

Receive Amplifier Subsystem
- RAS

Feed Array - FA

Patch Patch

Patch Patch

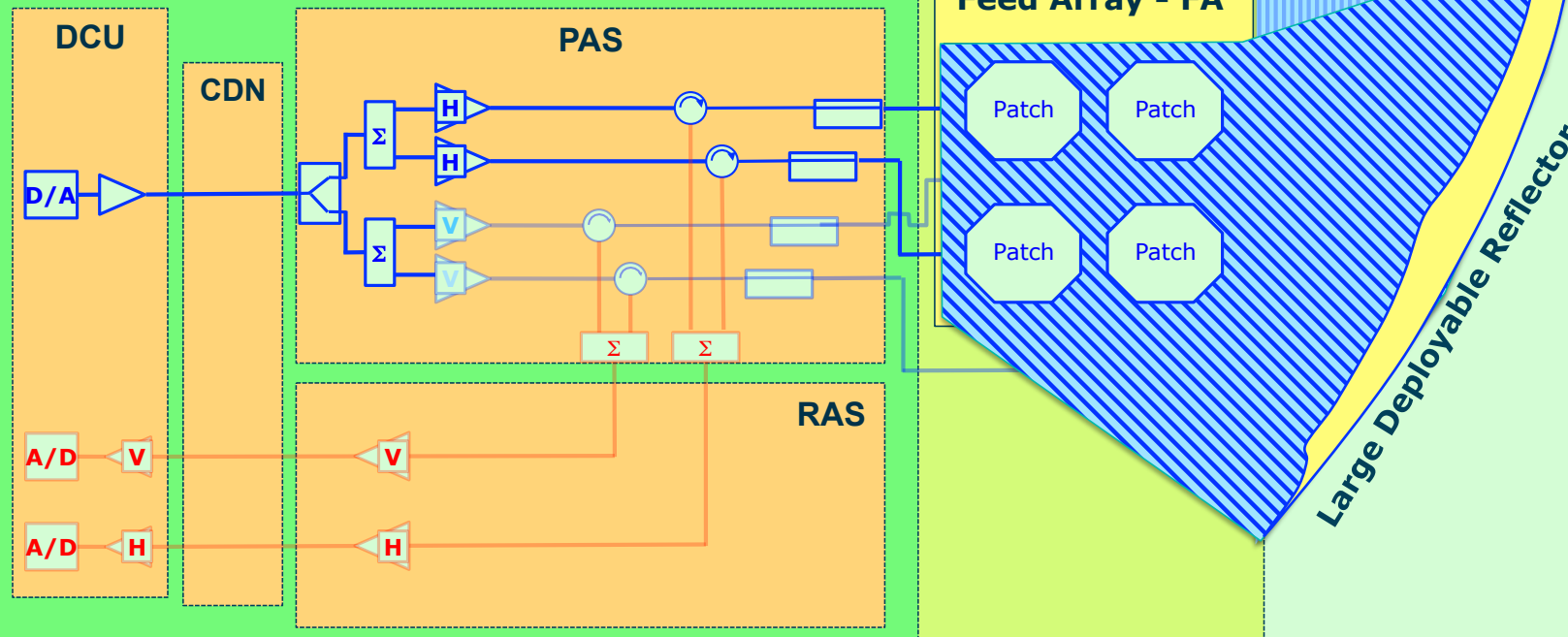


Biomass SAR Instrument: Full polarization

Payload

Instrument

INstrument Electronics System - INES



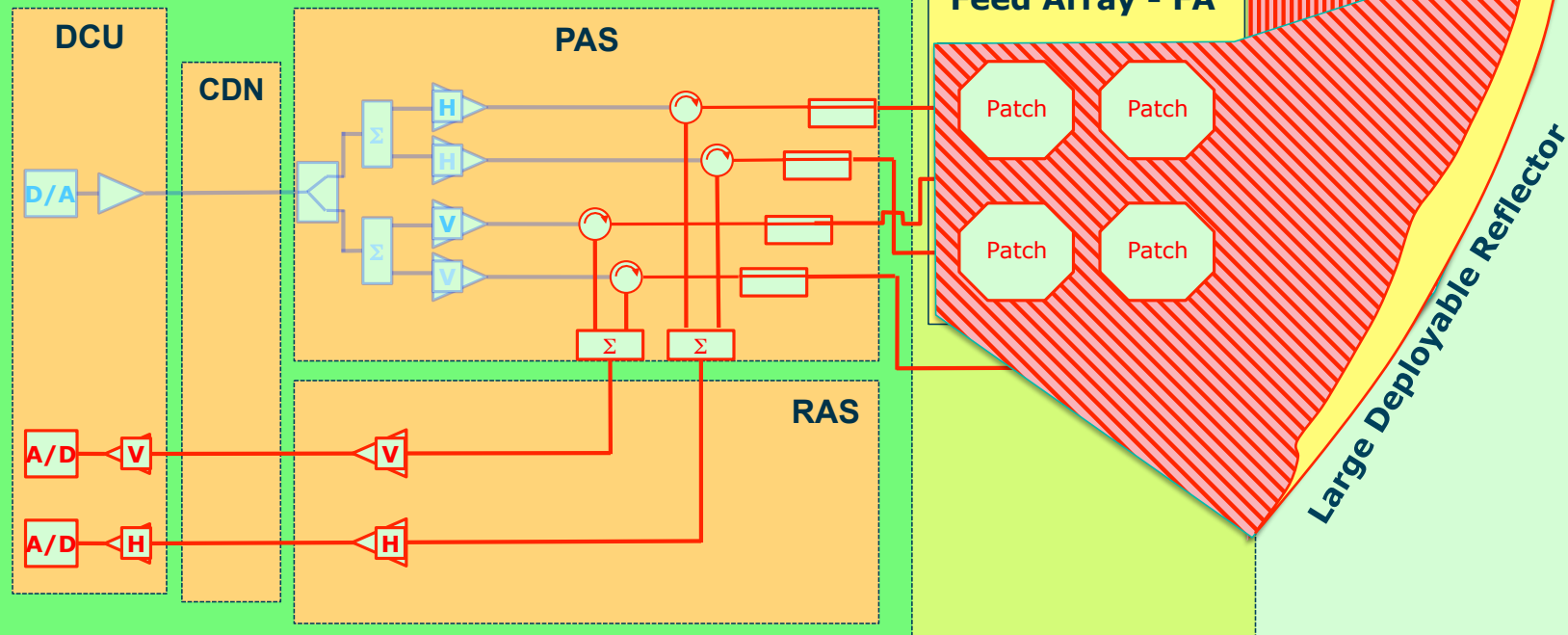
Biomass SAR Instrument: Full polarization



Payload

Instrument

INstrument Electronics System - INES

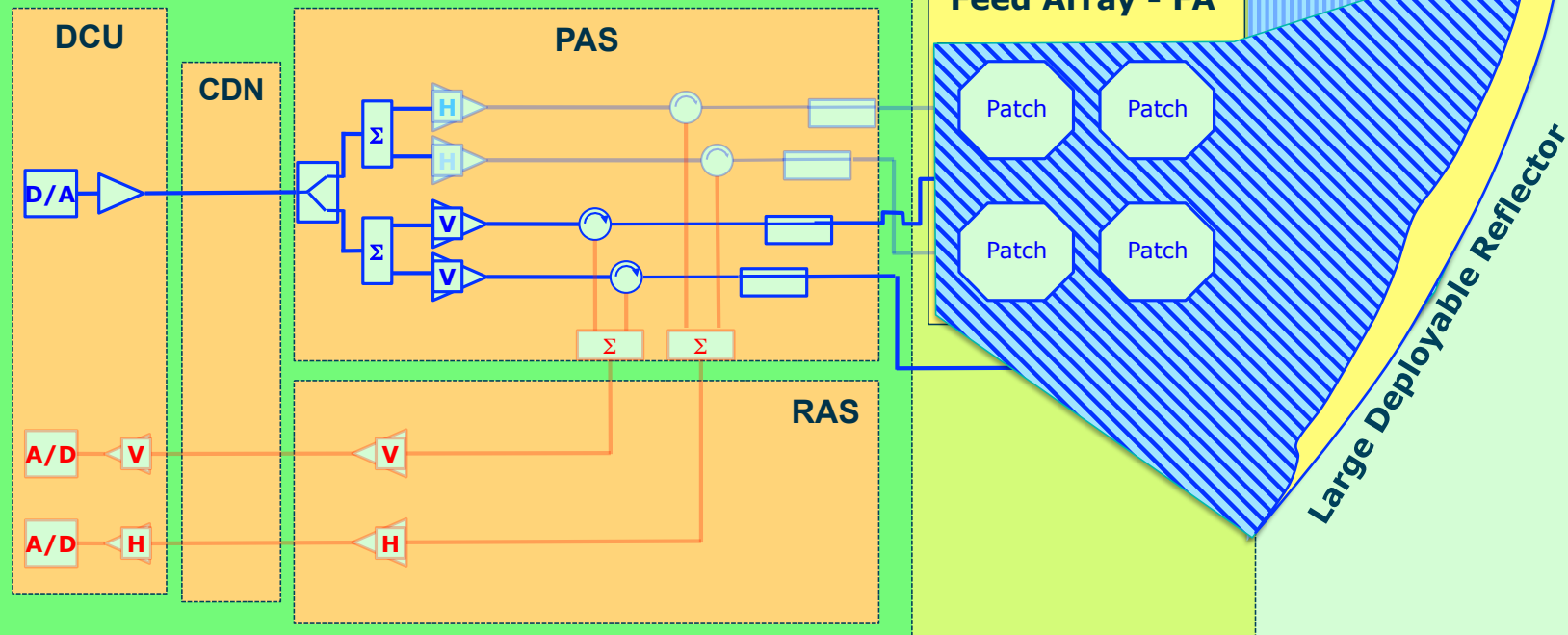


Biomass SAR Instrument: Full polarization

Payload

Instrument

INstrument Electronics System - INES

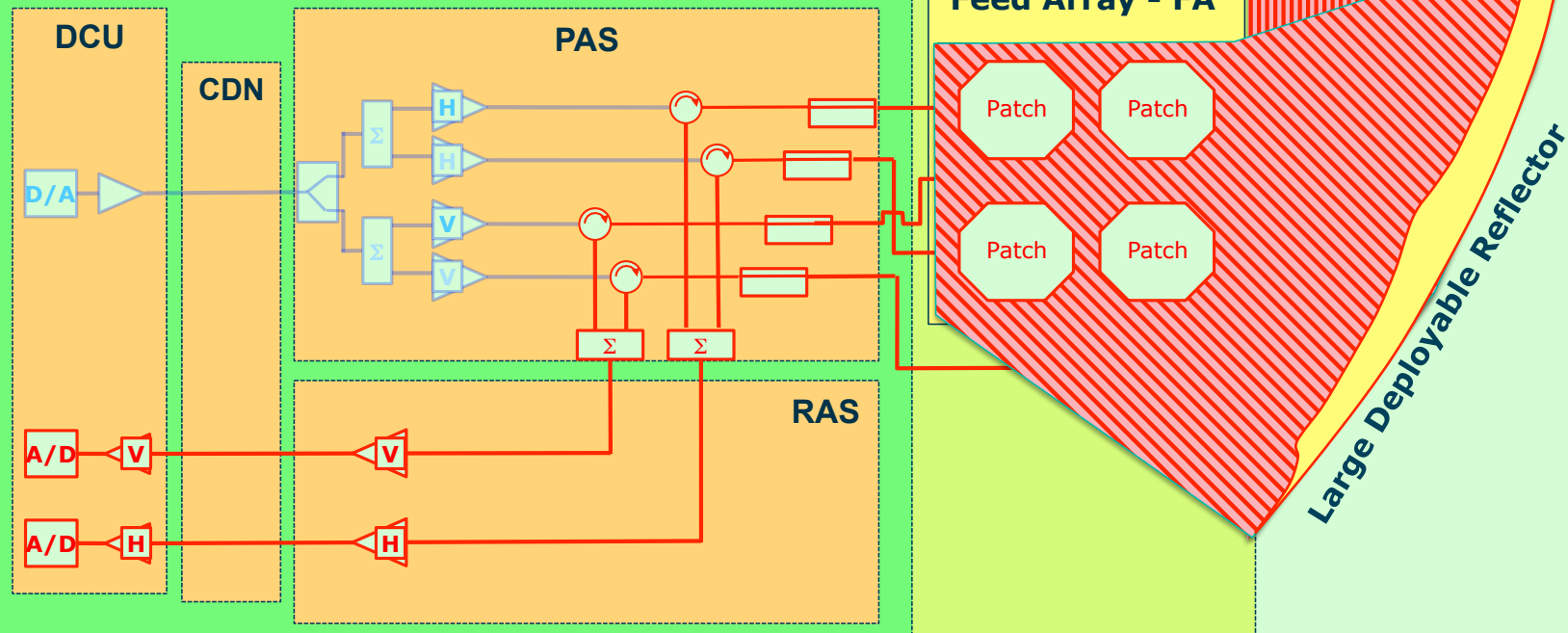


Biomass SAR Instrument: Full polarization

Payload

Instrument

INstrument Electronics System - INES



Biomass SAR Instrument: Internal calibration (ICAL)

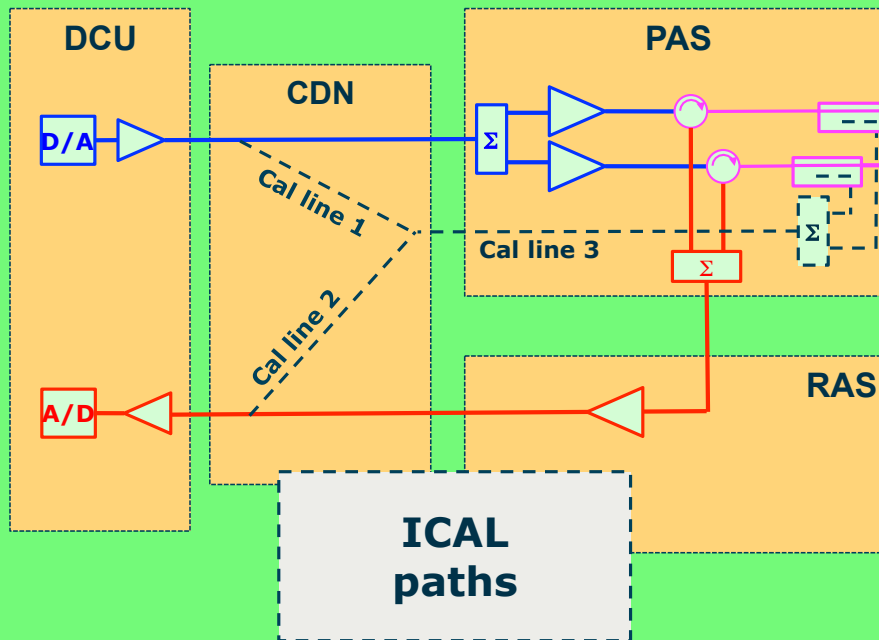
Payload Instrument

INstrument Electronics System - INES

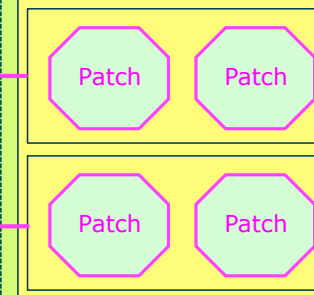
--- one polarization shown ---

chirp generation and Tx

Rx and sampling



Feed Array - FA



Biomass SAR Instrument: ICAL residual/Charact./Ext. calibration

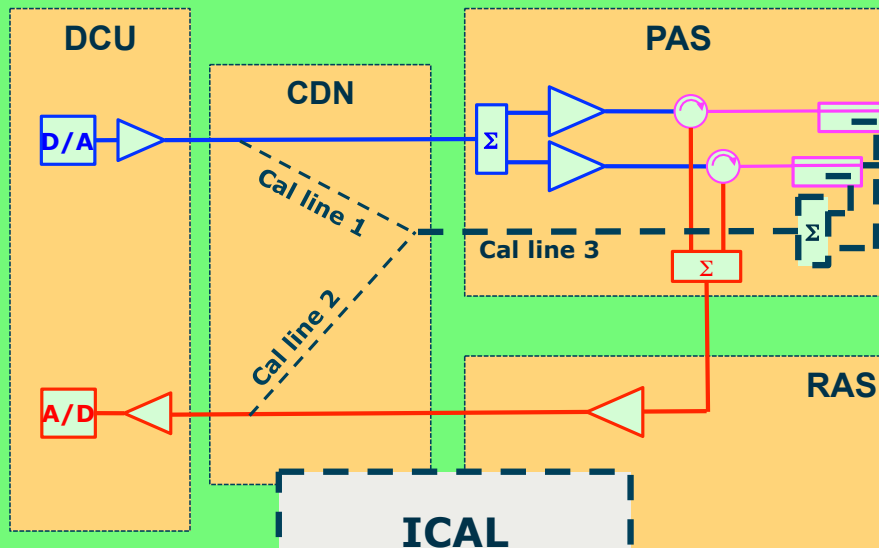
Payload Instrument

INstrument Electronics System - INES

--- one polarization shown ---

chirp generation and Tx

Rx and sampling



Tx/Rx not monitored by ICAL

Large Deployable Reflector



Biomass SAR Instrument: ICAL residual/Charact./Ext. calibration

Payload

Instrument

INstrument Electronics System - INES

--- one polarization shown ---

chirp generation and Tx

Rx and sampling

DCU

D/A

CDN

Cal line 1

Cal line 2

Earth Observation Systems are measurement systems (not telecom!), their key objective is to provide accurate measurements

Cal line 3

Σ

RAS

ICAL residuals

Patch

Patch

Tx/Rx

Tx/Rx not monitored by ICAL

Large Deployable Reflector



Biomass SAR Instrument: ICAL residual/Charact./Ext. calibration



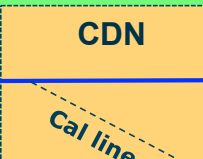
Payload

Instrument

INstrument Electronics System - INES

--- one polarization shown ---

chirp generation and Tx



Earth Observation Systems are measurement systems (not telecom!), their key objective is to provide accurate measurements

Tx/Rx not monitored by ICAL

Rx and sampling

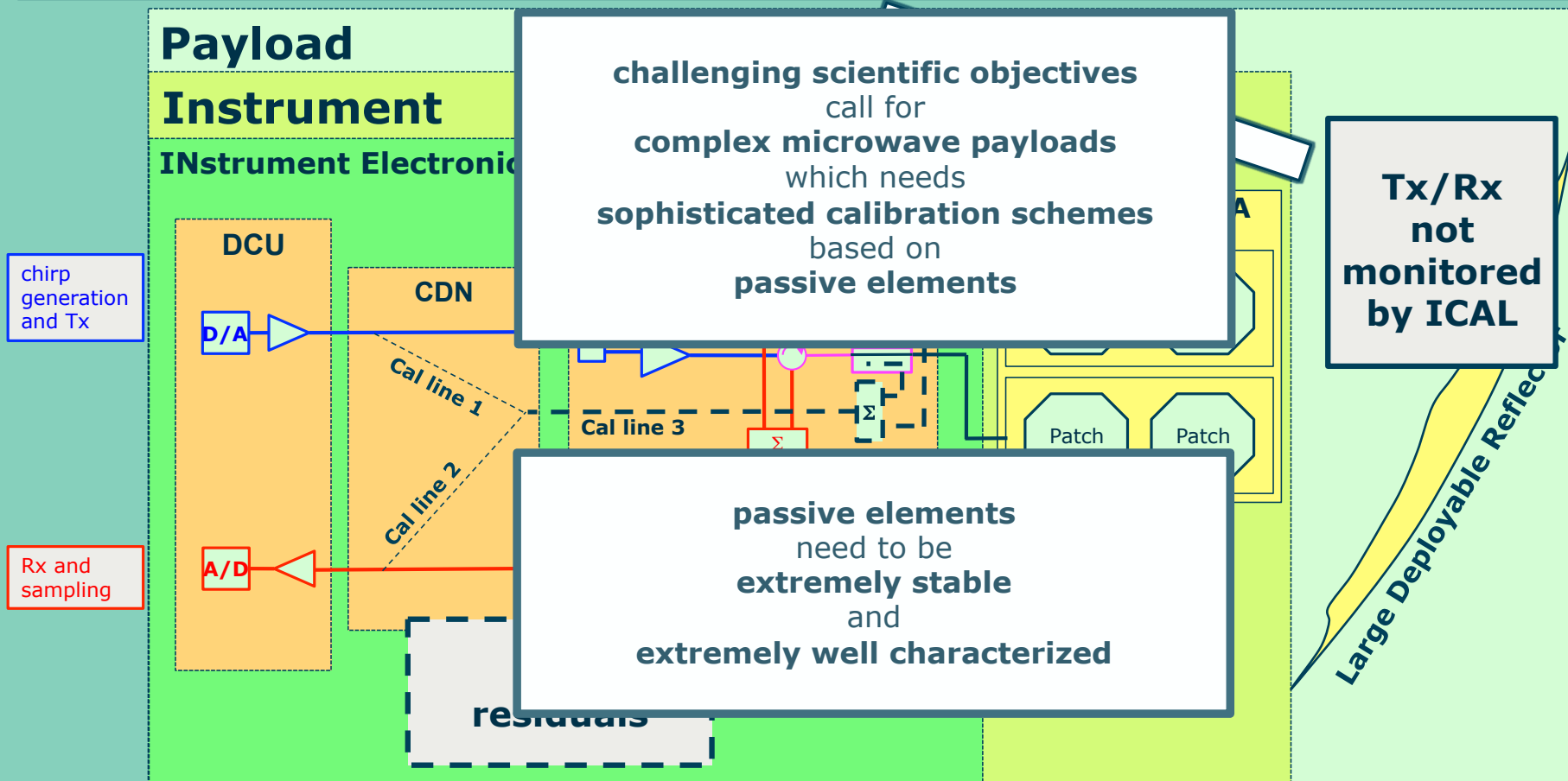
in cases, the accuracy of the scientific estimation is the accuracy of the on-ground characterization of the passive elements

ICAL residuals

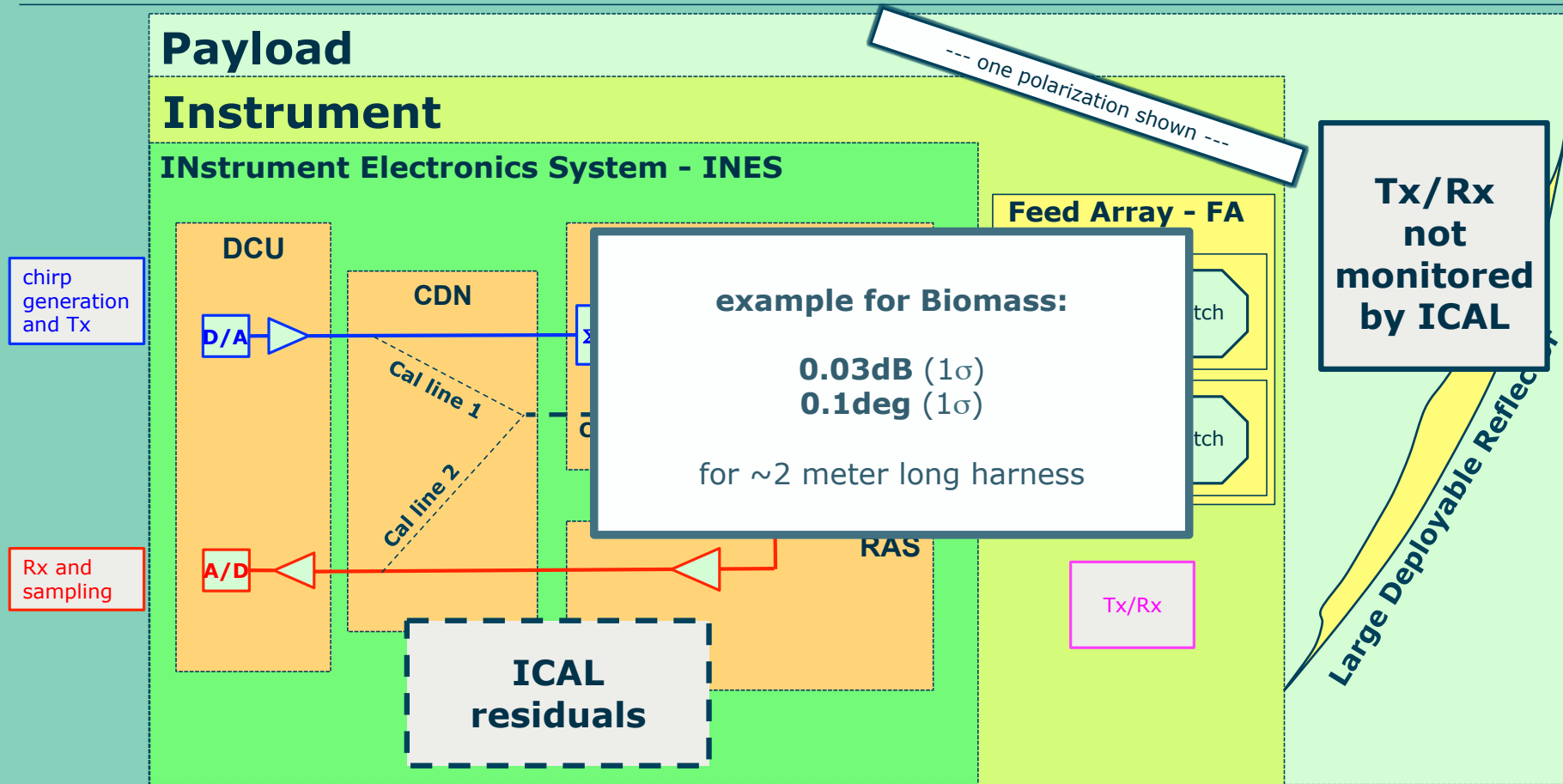
Large Scale Reflectivity



Biomass SAR Instrument: ICAL residual/Charact./Ext. calibration



Biomass SAR Instrument: ICAL residual/Charact./Ext. calibration



Payload

Instrument

INstrument Electronics System - INES

--- one polarization shown ---

Feed Array - FA

Tx/Rx

not stored
ICAL

able Reflect

chirp generation and Tx

DCU

D/A

in cases, the accuracy of the scientific estimations is the accuracy of the on-ground characterization of the passive elements

so, please

**ACCURATE CHARACTERIZATION,
ACCURATE CHARACTERIZATION,
ACCURATE CHARACTEIRZATION**

Rx and sampling

A/D

Outline

- Biomass mission, its objectives and its satellite ✓
- Biomass Payload ✓
- Payload calibration and the importance of the passive elements ✓

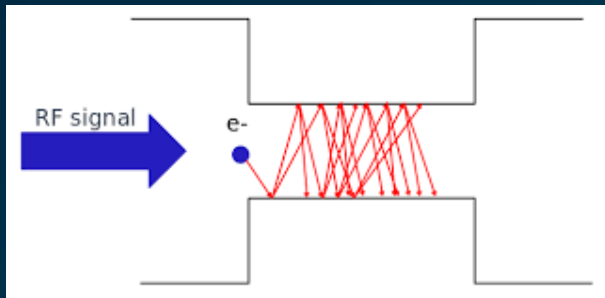
- **Multipaction: a special topic**
- **Some other passive EEE topics**

- Mission status



Multipactor Challenges

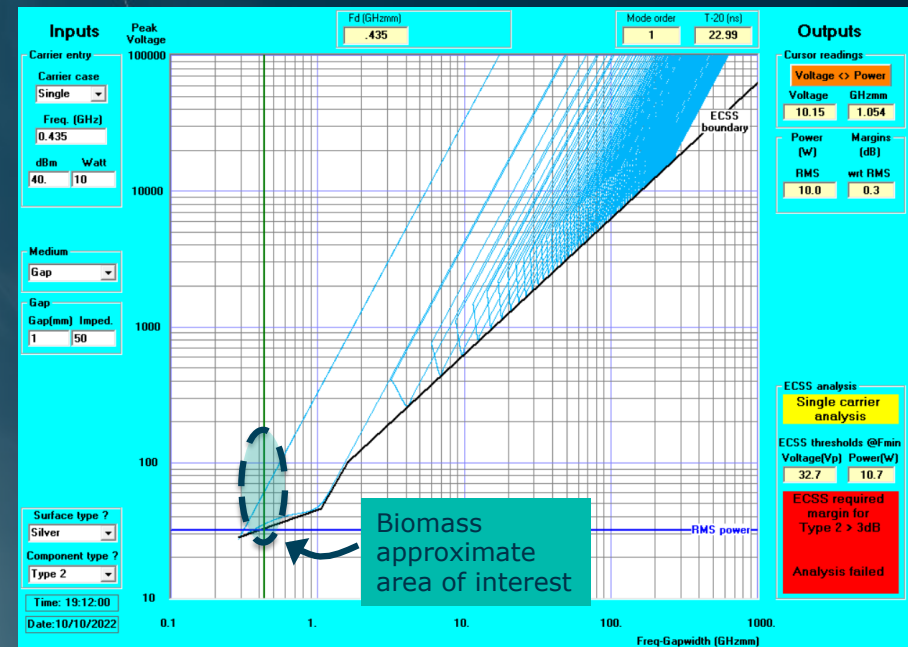
Multipactor effect is a RF breakdown in vacuum phenomenon, where, under certain conditions, secondary electron emission in resonance with an alternating electric field leads to exponential electron multiplication (electron avalanche), possibly damaging and even destroying the RF device.



The following parameters:

- Frequency = 435MHz
- Power = 10W
- Gap > 1mm

can theoretically trigger a multipaction discharge!



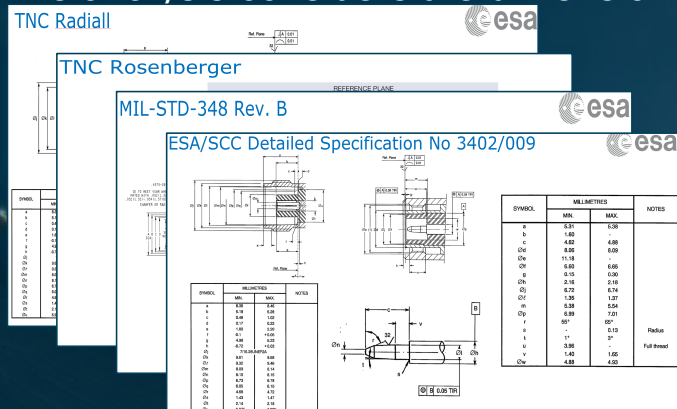
Multipactor Challenges: TNC Connectors

Each Payload subsystem is composed of a number of stand-alone RF modules equipped with coaxial connectors, then interconnected by RF coaxial cables.

There is a different supplier for each module/equipment. This implies a compatibility issue between the interface of the RF harness and the equipment/components.

A detailed verification by analysis of the compatibility with respect to multipactor discharge between the TNC connectors of the equipments/components and those of the RF coaxial cables has been carried out.

The analysis considers the dimensions and tolerances of the interfaces between different connectors'.



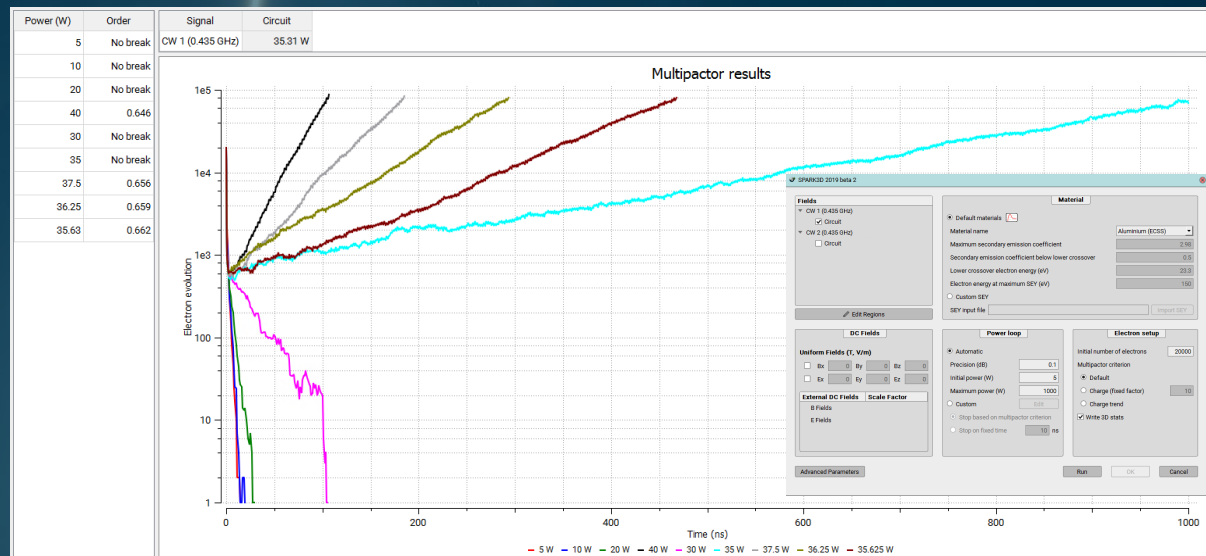
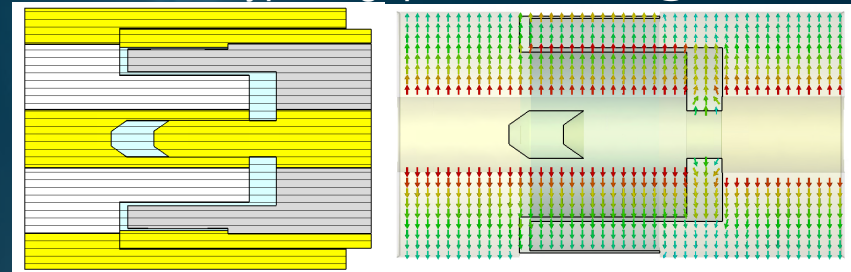
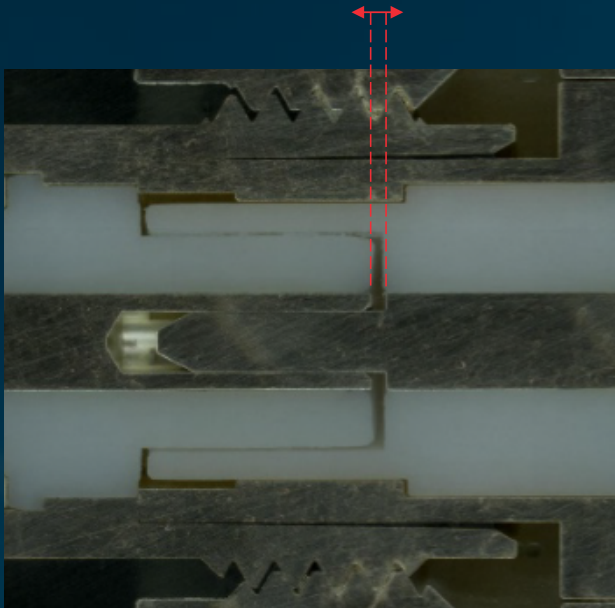
		ROSENBERGER		RADIALL		No 3402/009		MIL-STD-348		
		<i>Min</i>	<i>Max</i>	<i>Min</i>	<i>Max</i>	<i>Min</i>	<i>Max</i>	<i>Min</i>	<i>Max</i>	
Dimensions in Cross-section	Male	Ø Inner Connector	5.50	5.57	5.50	5.50	5.50	5.57	5.50	5.57
		Ø Internal PTFE	4.93		4.90	4.93	4.90	4.93	4.93	
	Ø External PTFE			6.60	6.67	6.60	6.67			
	Ø Internal Housing			6.72	6.78	6.72	6.78			
Female	Ø External PTFE		4.75	4.68	4.72	4.65	4.62		4.75	
	Ø Internal Housing	6.50	6.57	6.50	6.55	6.00	6.54	6.50	6.57	
Longitudinal Dimensions	Male	Depth PTFE wrt Ref	5.58		5.58	5.54	5.58	5.54	5.58	5.75
		Depth Ref to PTFE			0.25	0.30	0.25	0.30	0.25	0.30
Female	Length PTFE wrt Ref	4.55	5.25	5.58	5.58	5.58	5.28	4.78	5.28	



Multipactor Challenges: TNC Connectors



SPARK 3D simulations show breakdown starting from 35W for a typical gap of ~0.7 mm @435MHz:

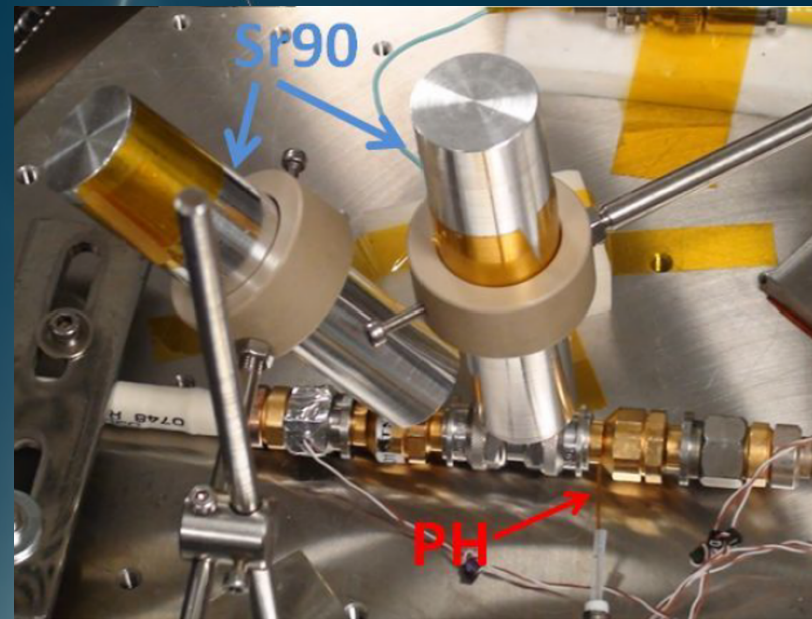
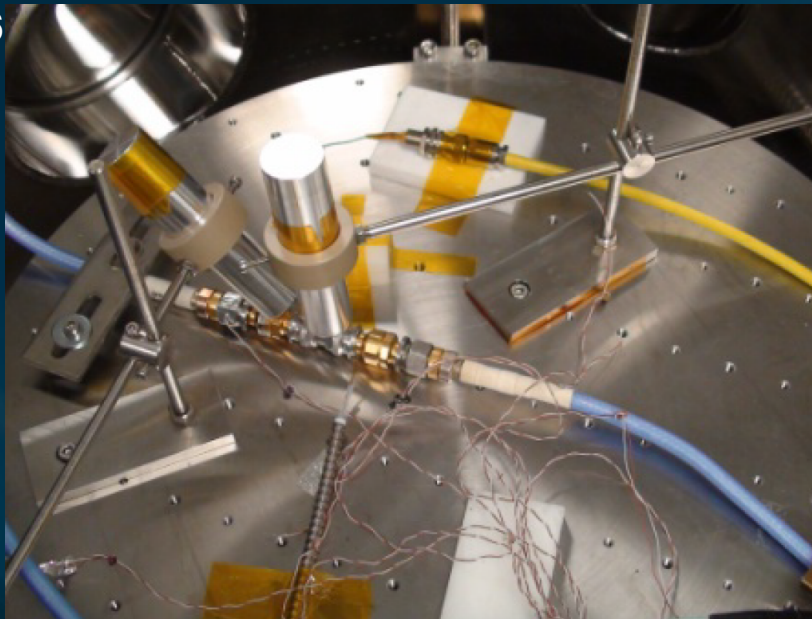


Multipactor Challenges: TNC Connectors

First tests show multipaction at 25W.

After shimming between connectors from different manufacturers: no threshold crossings were seen up to

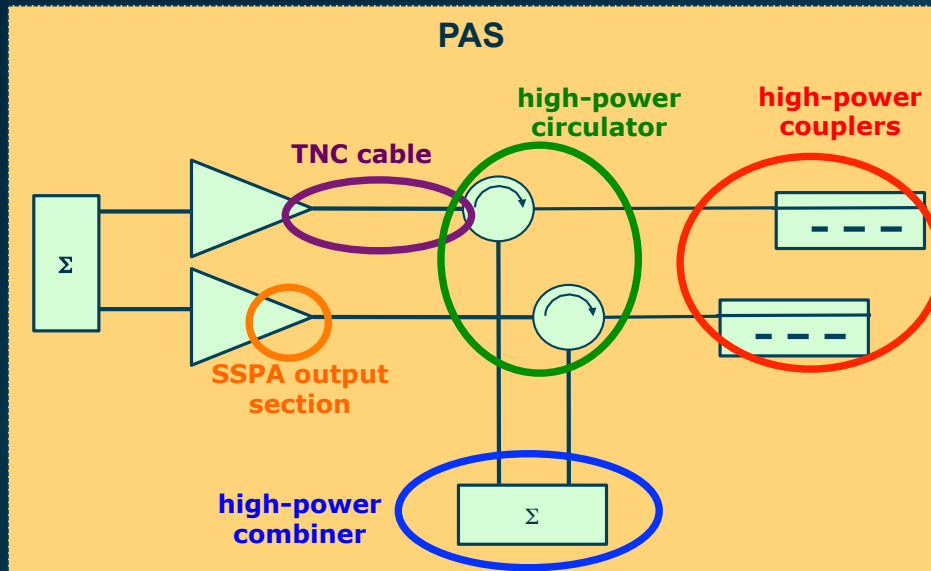
6



435 MHz | no Discharges up to at least 600 W

Multipactor Challenges: PAS RF Passive Components

The PAS subsystem is the subsystem that handles and transmit the RF high power to the **antenna**

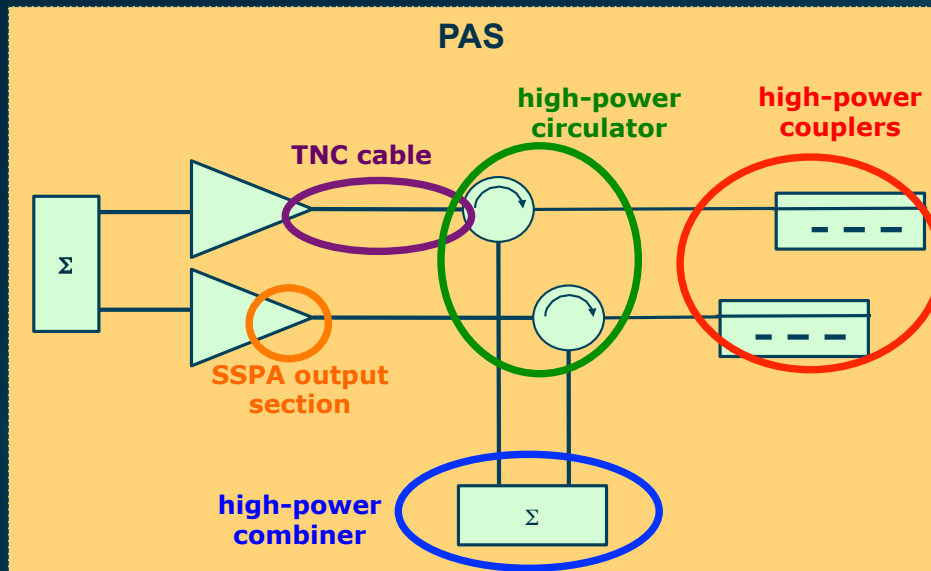


Note: RF switch was treated as a self-standing equipment so multipaction was treated at that level (within the qualification of the equipment)

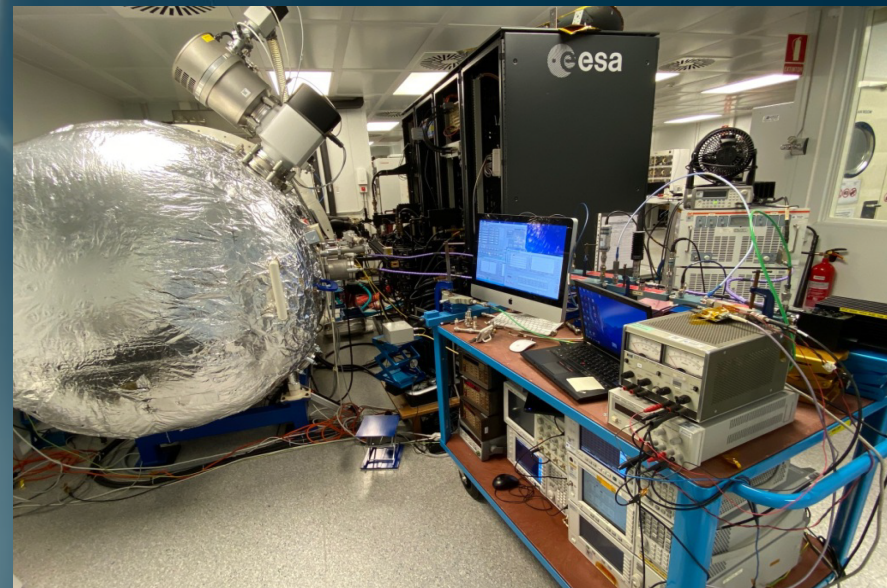
Multipactor Challenges: PAS RF Passive Components



The PAS subsystem is the subsystem that handles and transmit the RF high power to the **antenna**



Note: RF switch was treated as a self-standing equipment so multipaction was treated at that level (within the qualification of the equipment)



Multipaction tests were conducted at the European High Power Laboratory in Valencia (Spain) on the following critical RF Passive parts:

(see next slide)



Multipactor Challenges: PAS RF Passive Components



High Power Combiner (Cobham)

High Power Coupler (Cobham)

High Power Coaxial Circulator (Cobham)

RF TNC Cable Assemblies (Times Microwave)

SSPA Output section (LND & Cobham)

Frequency = 435 MHz – RF Power up to ~800Wp

Temperature = -30C +60C

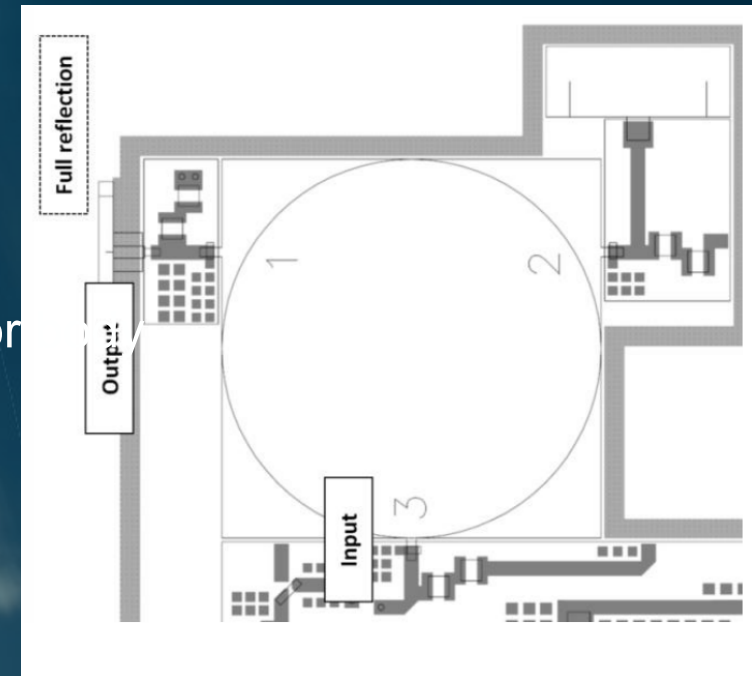
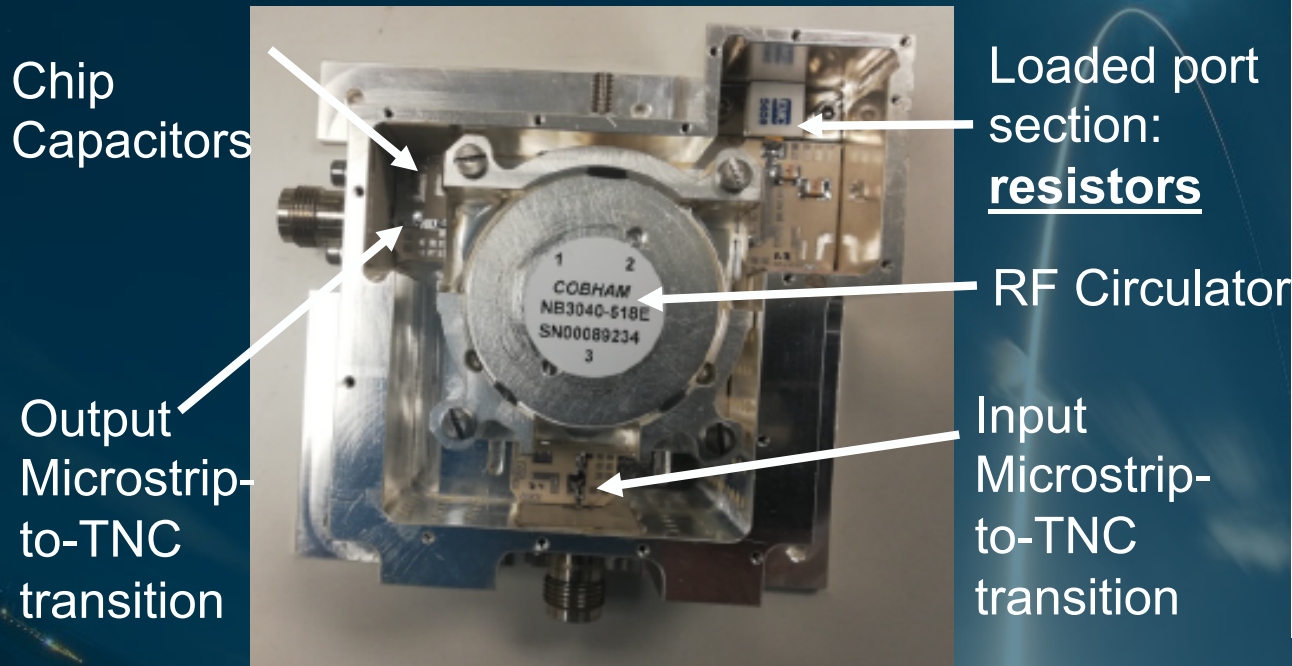
Different conditions:

- Full reflection and Matched output with maximum level for the Coupler
- Forward configuration for the Combiner
- Forward and Full reflection for the Circulator
- Forward for the RF Cable Assemblies
- Forward and Full reflection for the SSPA Output section



Multipactor Tests: a specific example

SSPA Output power section (LND & Cobham)

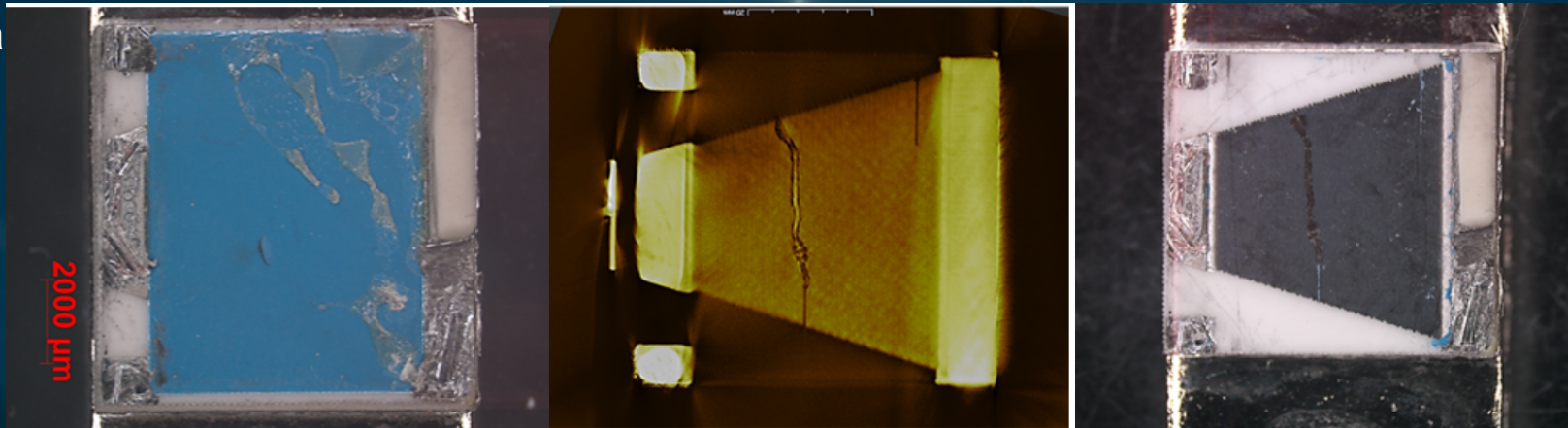


It was detected one discharge after 15 minutes at 630 W peak. Both local and global methods were triggered. After this anomaly, the return losses suffered degradation to around 3dB. Measurement of the commercial load shows an open circuit (9K Ohm!).

Multipactor Challenges: a resistor multipaction

- The electrical arcing has been produced at the end of the trimming line of the internal impedance transformer adaptation of the commercial load.
- The resistance has dropped locally at this end point due to an increase in temperature, this implied an electrical arcing, with a concentration of the arc fault current and voltage at one place (the end of the trimming line). The arc has then propagated to the nearest ground and crossed and damaged the resistive layer.

• The ca
power.



Multipaction: Conclusions & Lessons Learned



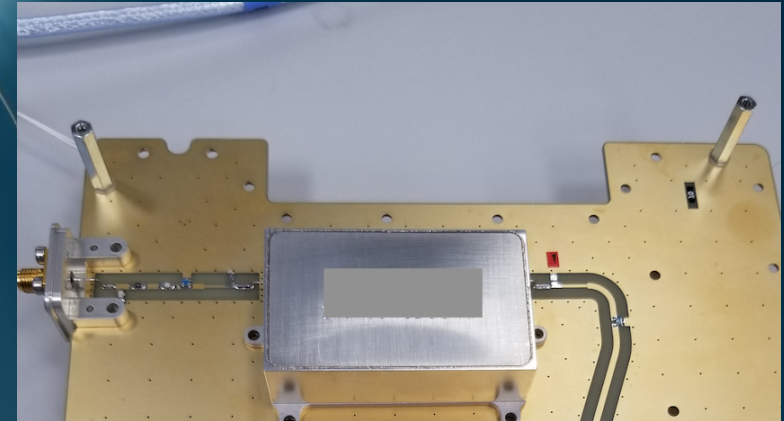
- The “claimed” heritage was not valid, mainly due to the specific requirements of BIOMASS, mainly low frequency (P-band) and High Power (~200W).
- Multipaction Analysis were giving a certain confidence on some parts. However, proper and representative (including flight model quality) Multipactor tests were performed as per BIOMASS’s requirements and including assembly/mounting conditions (e.g. SMT/microstrip parts, torque of connectors, etc.)
- Adding Teflon shim (suitable gasket) sheets between connectors helps to fill the gaps and reduce risk of multipaction discharges.
- Internal filling with a proper potting in the internal designs of RF parts helps fills the gaps, as well as adding proper coating on SMT/microstrip interfaces.

• Qualification tests of the payload instrument equipment show good results!



RF filters

Challenging activity: difficult qualification/acceptance



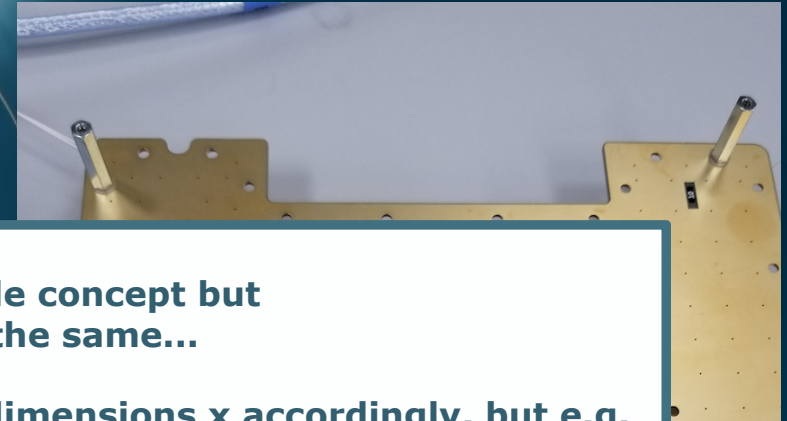
Several actions were taken and lesson learned:

- Improve the internal design and to procure such filters as RF filters (or, even as equipment!) instead of hybrids.
- Adapt the LAT (Group C) testing to the project requirements, flight representative test flow with margin (i.e. thermal range, constant acceleration down to e.g. 250g).
- Add CSAM inspections (in-process control, screening and LAT) to be able to properly screen the parts and check the evolution of internal soldering/attachment and voids.

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Challenging activity: difficult qualification/acceptance



Note: scaling is a valuable concept but not everything scale the same...

adapting for longer wavelength -->> scale dimensions x accordingly, but e.g.

Several actions

**surface areas scale with the square: x^2
weight scale with the cube: x^3 !**

instead of

- Improve the hybrids.
- Adapt the LAT (Group C) testing to the project requirements, flight representative test flow with margin (i.e. thermal range, constant acceleration down to e.g. 250g).
- Add CSAM inspections (in-process control, screening and LAT) to be able to properly screen the parts and check the evolution of internal soldering/attachment and voids.

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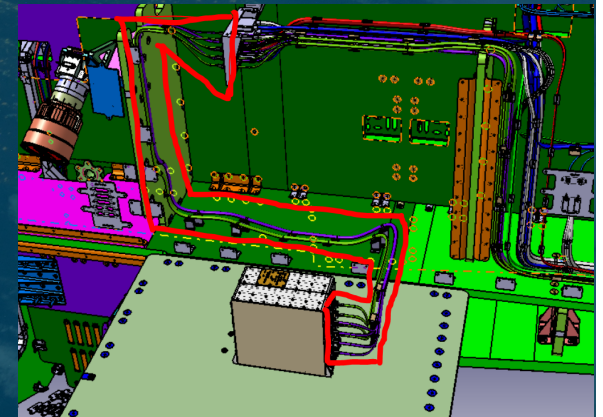
S/C Flight Harness + Payload Instrument Harness



DC, Power, RF and Wizardlink Harness

Procurement of ESCC QPL components (or according to):
ESCC 3401, ESCC3402, ESCC3901, ESCC3902, ESCC3408

Project qualification:
Axowave (AXON, FR) for Wizardlink (high data transfer).
+ SiO₂ RF phase-stable semi-rigid (Times Microwave)

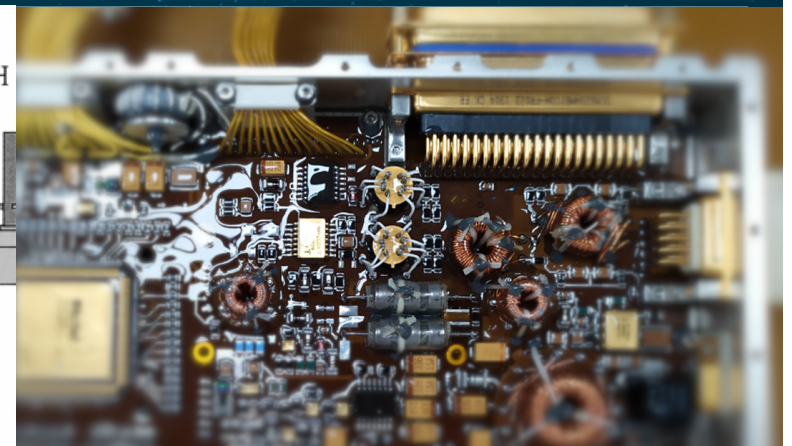
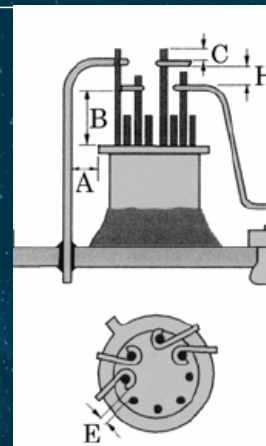


SMT related issues: Relays upside/down

- Relays (as well as oscillators, switches, crystals) are considered as sensitive parts to mechanical loads.

- TO-5 relay has been mounted upside/down!

- This Relay is QPL but has not been tested in this specific mounting conditions.



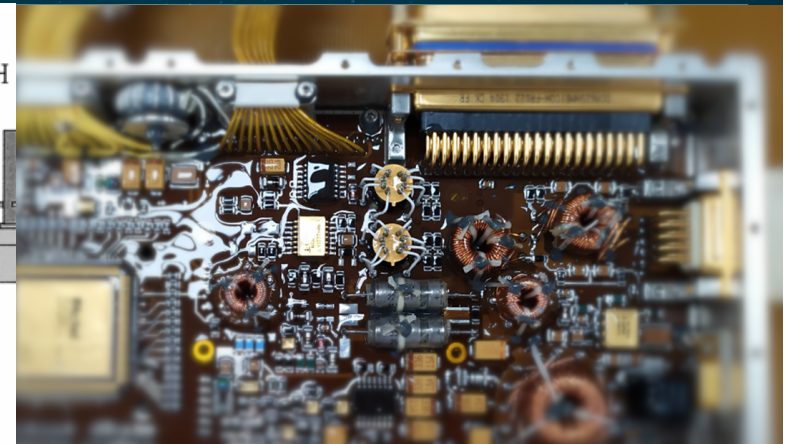
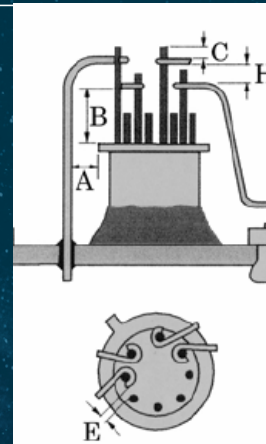
- At the time of the procurement, ECSS-Q-ST-70-08 allowed this specific mounting for TO-5 package, **without** any verification assembly (i.e. 500 thermal cycling, vibration and DPA).

- ESA Recommendation was to :

1. Verify that the relay has been procured properly, i.e. QPL or with LAT testing including vibration and shock on the three axes,
2. Ensure QM have been mechanically tested and relays have successfully switched afterwards,
3. Use a less rigid adhesive, i.e. flexible adhesive based on Silicone instead of rigid adhesive.

SMT related issues: Relays upside/down

- Relays (as well as oscillators, switches, crystals) are considered as sensitive parts to mechanical loads.
- TO-5 relay has been mounted upside/down!
- This Relay is QPL but has not been tested in this specific mounting conditions.
- ECSS-Q-ST-70-08C Standard has been **cancelled and superseded** by **ECSS-Q-ST-70-61C**, since April 2022. This new standard do NOT allow this specific mounting configuration for components sensitive to mechanical loads, unless a proper verification assembly campaign has been conducted.




ECSS-Q-ST-70-61_1510331

Components sensitive to vibrations and shock tests shall not be assembled as described in requirement 8.2.6a.

NOTE Examples of such sensitive components are: relays, oscillators, crystals. ECSS-E-HB-32-25A is providing a list of components known as sensitive to vibration and shock.

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- Multipaction: a special topic ✓
 - Some other passive EEE topics ✓
- **Mission status**

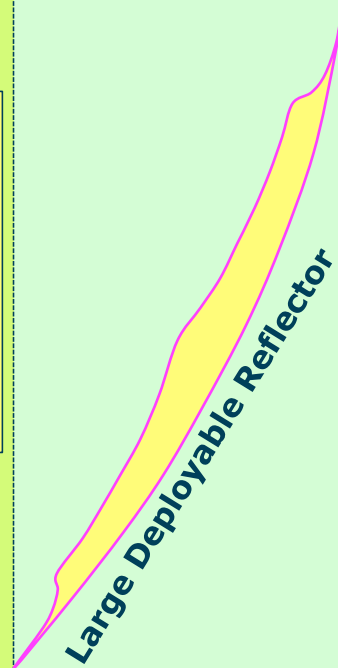
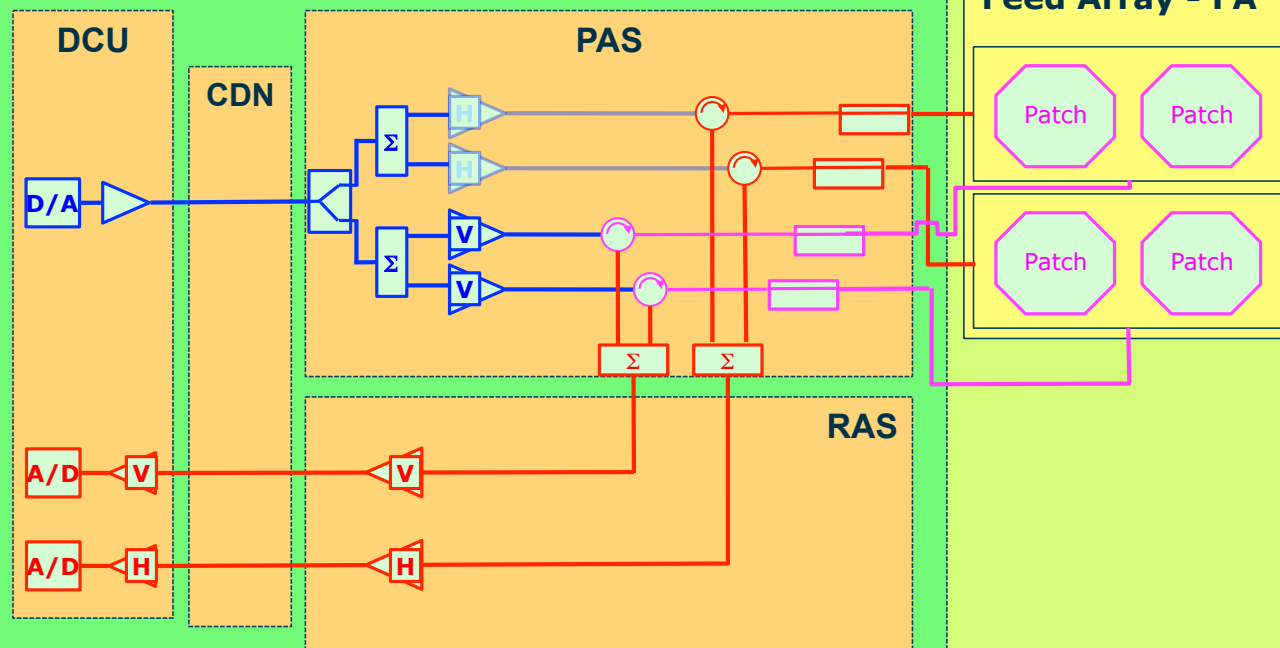
Biomass SAR Instrument: flight HW being delivered



Payload

Instrument Prime

INstrument Electronics System - INES



Biomass SAR Instrument: flight HW being delivered

Payload

Instrument Prime

INstrument Ele

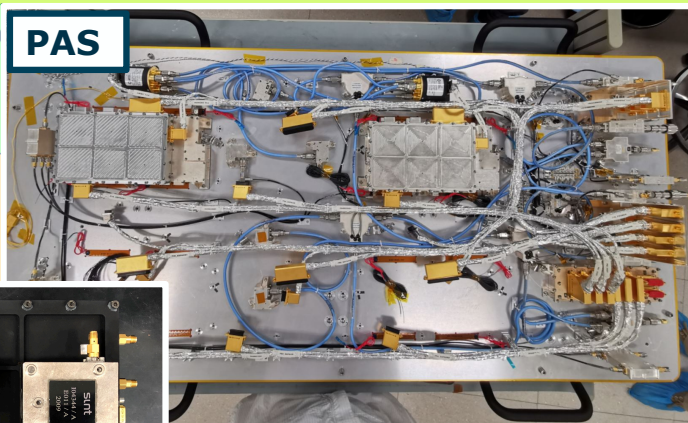
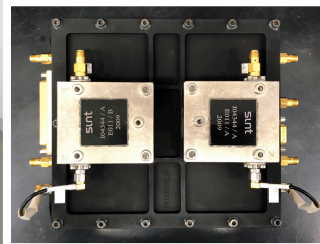
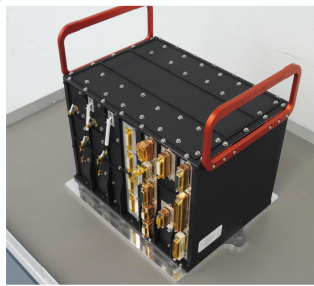
PAS

DCU

CDN

DCU

CDN

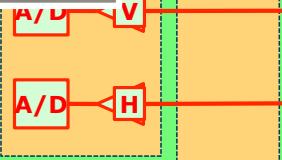
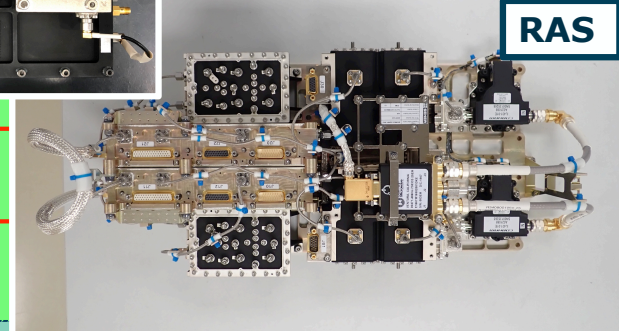


Feed Array - FA

Feed Array



RAS



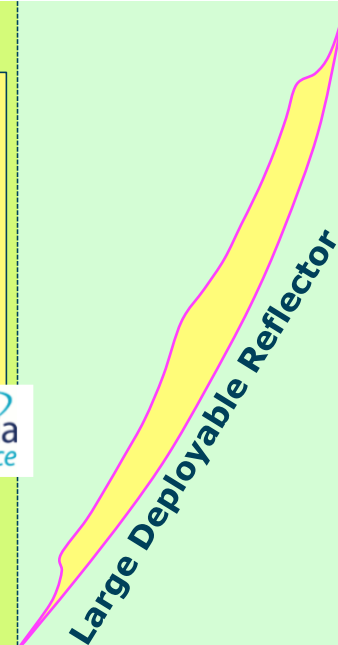
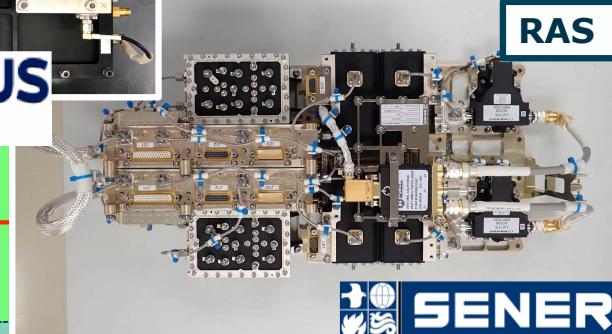
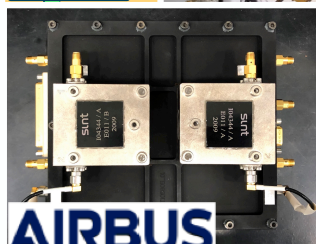
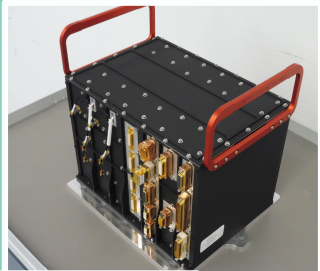
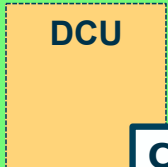
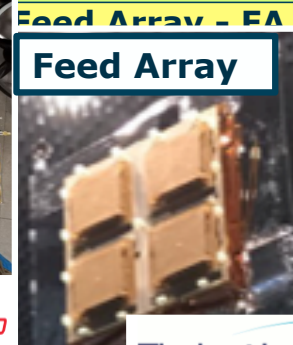
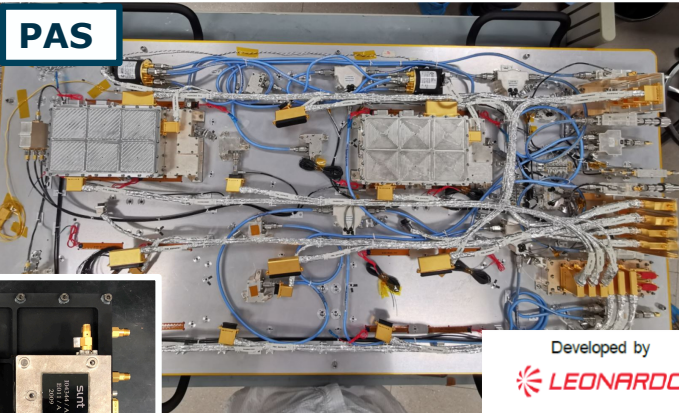
Biomass SAR Instrument: European consortium led by Airbus Germany



Payload

Instrument Prime AIRBUS Airbus Defence and Space GmbH, Germany

INstrument Ele



Biomass: Space segment

AIRBUS

Airbus Defence and Space Ltd, UK sa



courtesy of Airbus Defence and Space Ltd.



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Biomass: Mission Status

AIRBUS

Airbus Defence and Space Ltd, UK **sa**



- **Q3/4 2022:** satellite integrated with reflector and Instrument Hybrid → Satellite to mechanical testing
- **2023:** completion of Payload PFM → finalization of satellite tests

Q2 2024: launch!



courtesy of Airbus Defence and Space Ltd.



→ THE EUROPEAN SPACE AGENCY

more info on Biomass

- **MISSION** → *Report for Mission Selection: Biomass* (ESA SP-1324/1, May 2012), European Space Agency, The Netherlands
- **BIOMASS SYSTEM** → Sedehi, M. et al.: Biomass - A fully polarimetric P-band SAR ESA mission, EUSAR 2020/2021
- **INSTRUMENT HW** → Fügen, T. et al.: Development Status of the Biomass SAR Instrument, EUSAR 2020/2021
- **SPACECRAFT** → Warren, C. et al.: Overview of the Biomass satellite, Proceedings of the IGARSS 2018, July 2018
- **general news** → <https://www.bbc.com/news/science-environment-62650129> (23/08/2022)



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here goes video

biomass thanks you very
much!

→ ESA'S FOREST MISSION

